



LituanicaSAT-1: lessons learned from the first Lithuanian satellite mission

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- Background story
- Project Management
- System Design
- Mission Operations
- Lessons learned
- LituanicaSAT-2

Background story



Photo Credit: V. Buzas

Background story

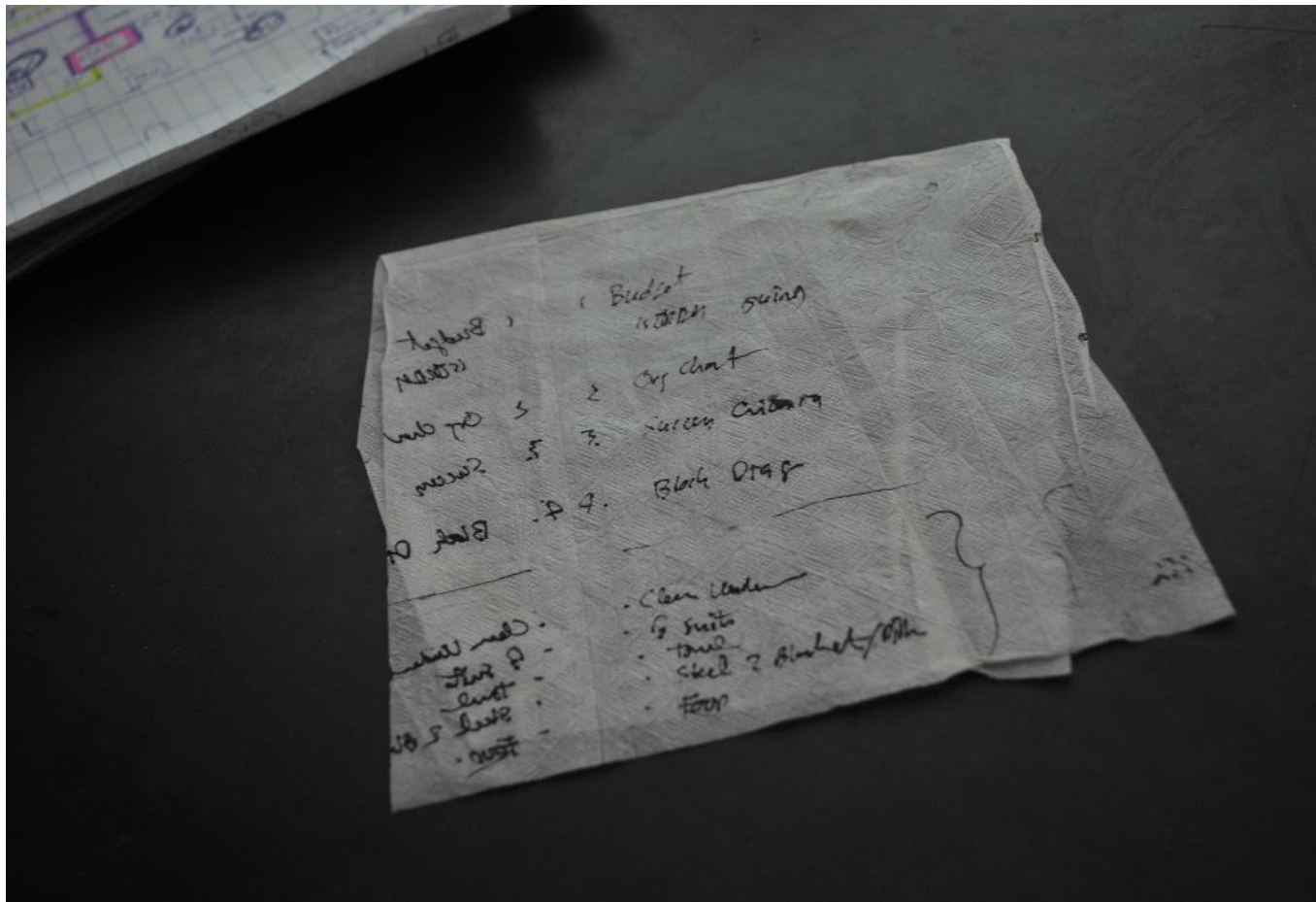
Our secret ingredients for a successful mission:

- **Work hard**
- **Do teambuilding**
- **Have passion in what you are doing**



Background story

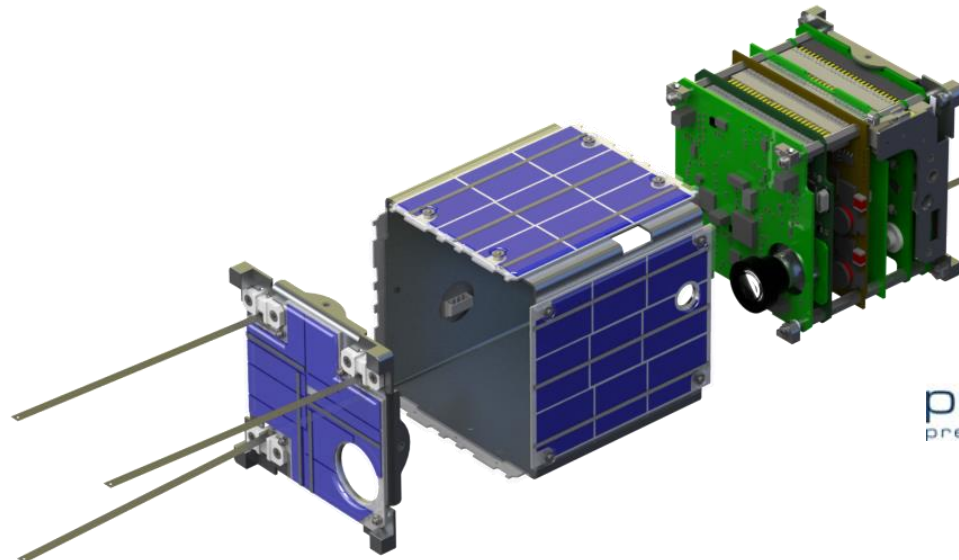
...and don't forget to use coffee brakes wisely



Mission Statement (or plan on nampkin)

- **To build and launch first Lithuanian satellite** of 1U standard CubeSat size
- To develop infrastructure and gain know how joining industry and academia
- To honour the **80th Anniversary** (July 15, 2013) of the flight across the Atlantic by Lithuanian pilots Steponas Darius and Stasys Girėnas.

Project Management



➤ Primary Mission Objective

- To build and launch a first Lithuanian 1U size cubesat and send Lithuania's first message from space

➤ Secondary Mission Objectives

- To provide university students and young engineers knowledge & real hands-on experience in satellite engineering
- To develop and test in space cubesat on board control and data handling sub-system
- To take pictures using on-board camera and downlink to the ground station
- To test an amateur radio FM voice repeater.

➤ **Minimum Success Criteria**

- NASA PSRP approval and delivery of the cubesat to the ISS
- Launch cubesat from ISS by the JEM Remote Manipulator System (JEMRMS)
- System initialization and antenna deployment after time-out
- Transmit telemetry data (RF Morse code beacon or packet radio)

➤ **Nominal Success Criteria**

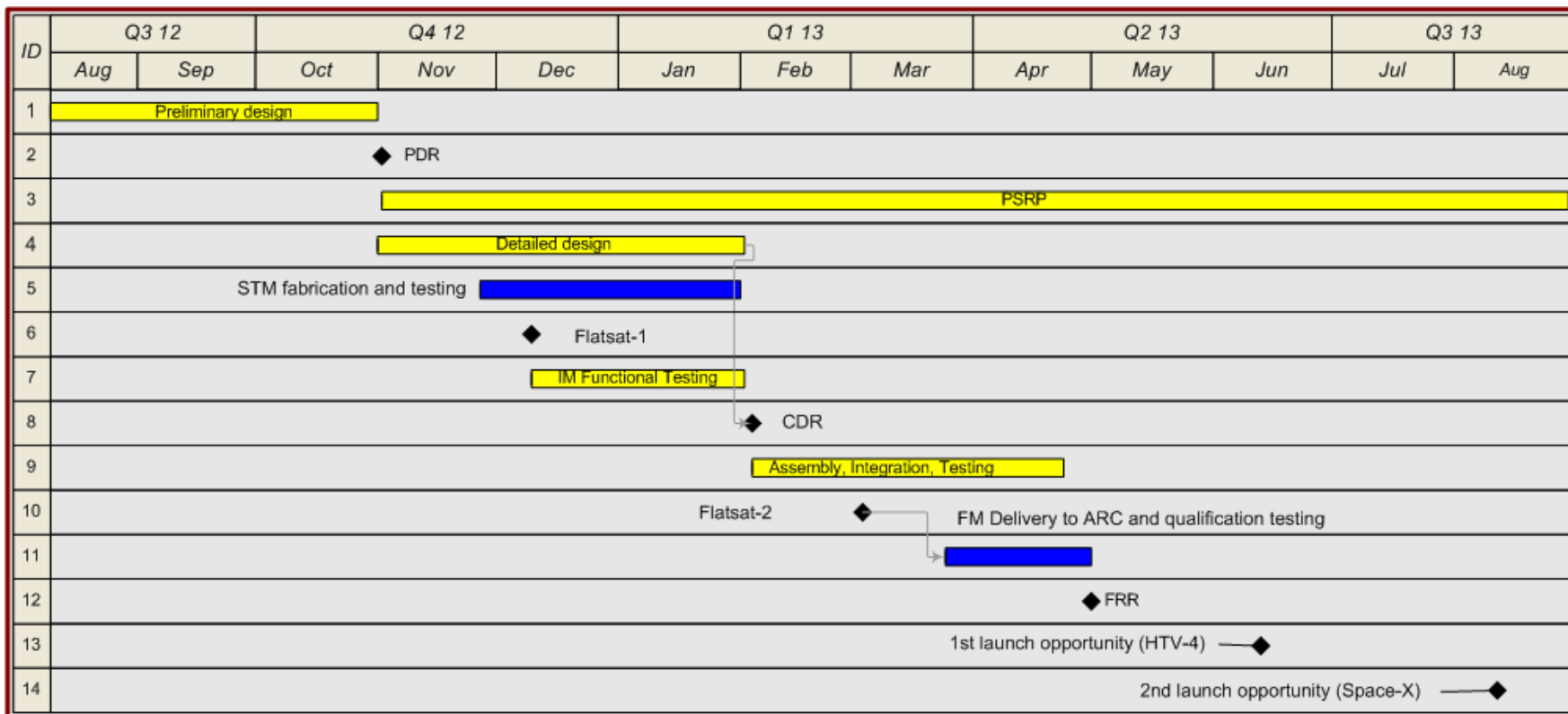
- Establish two-way communication with ground station
- Turn on the FM voice repeater

➤ **Comprehensive Success Criteria**

- Take pictures with camera and downlink data to the ground station

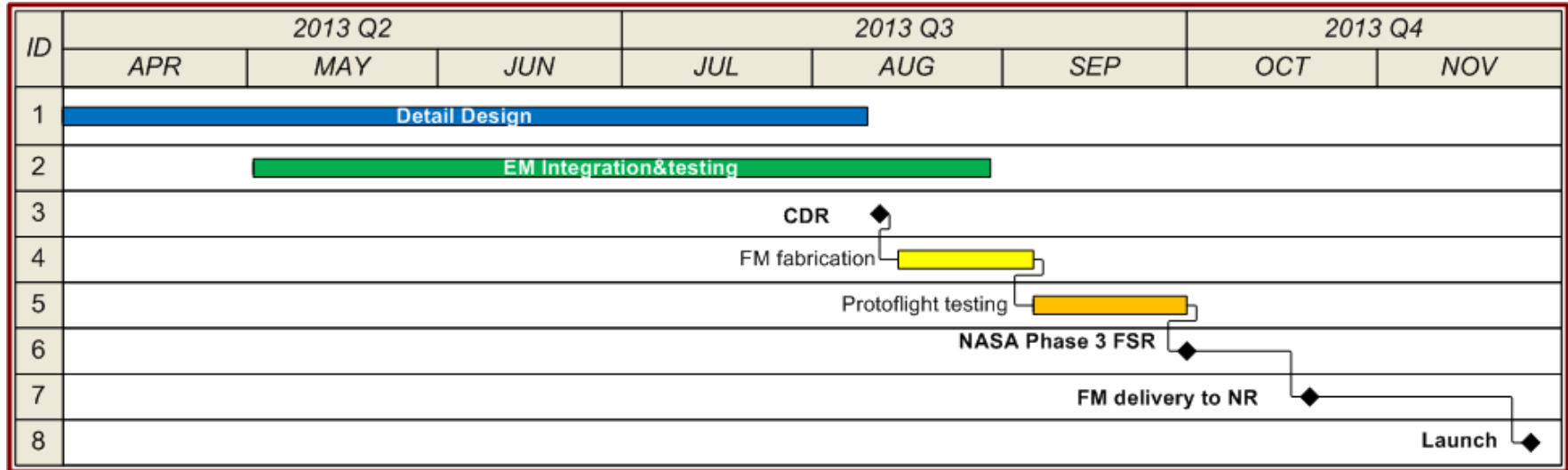
Schedule – Baseline 1

Very optimistic!



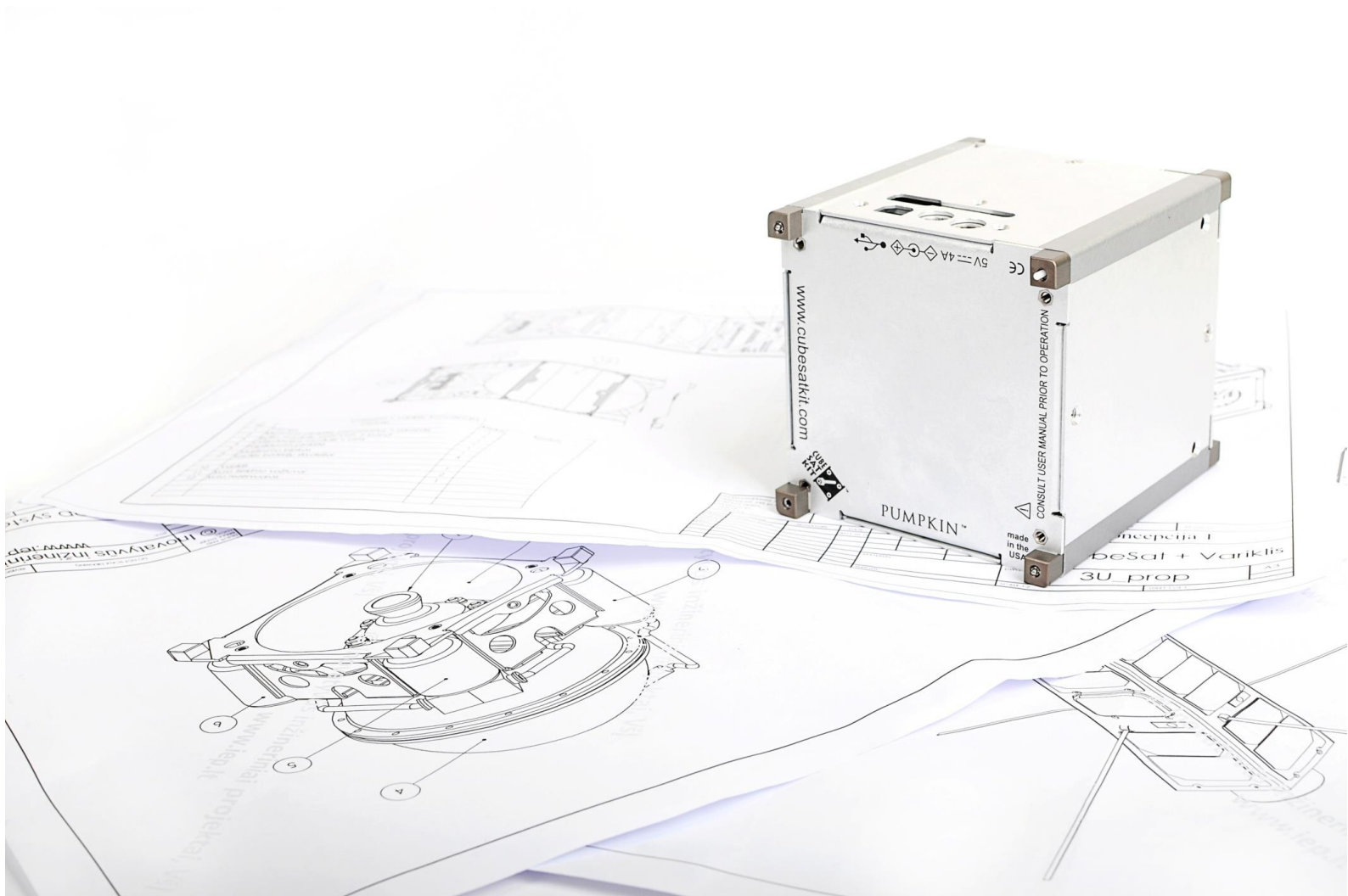
Schedule – Baseline 2

More realistic but still tight...

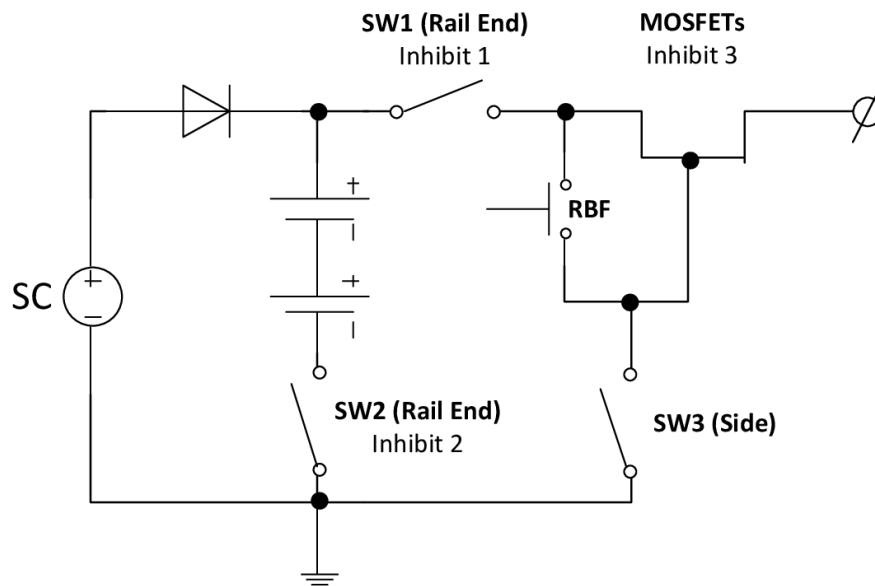


Legend:

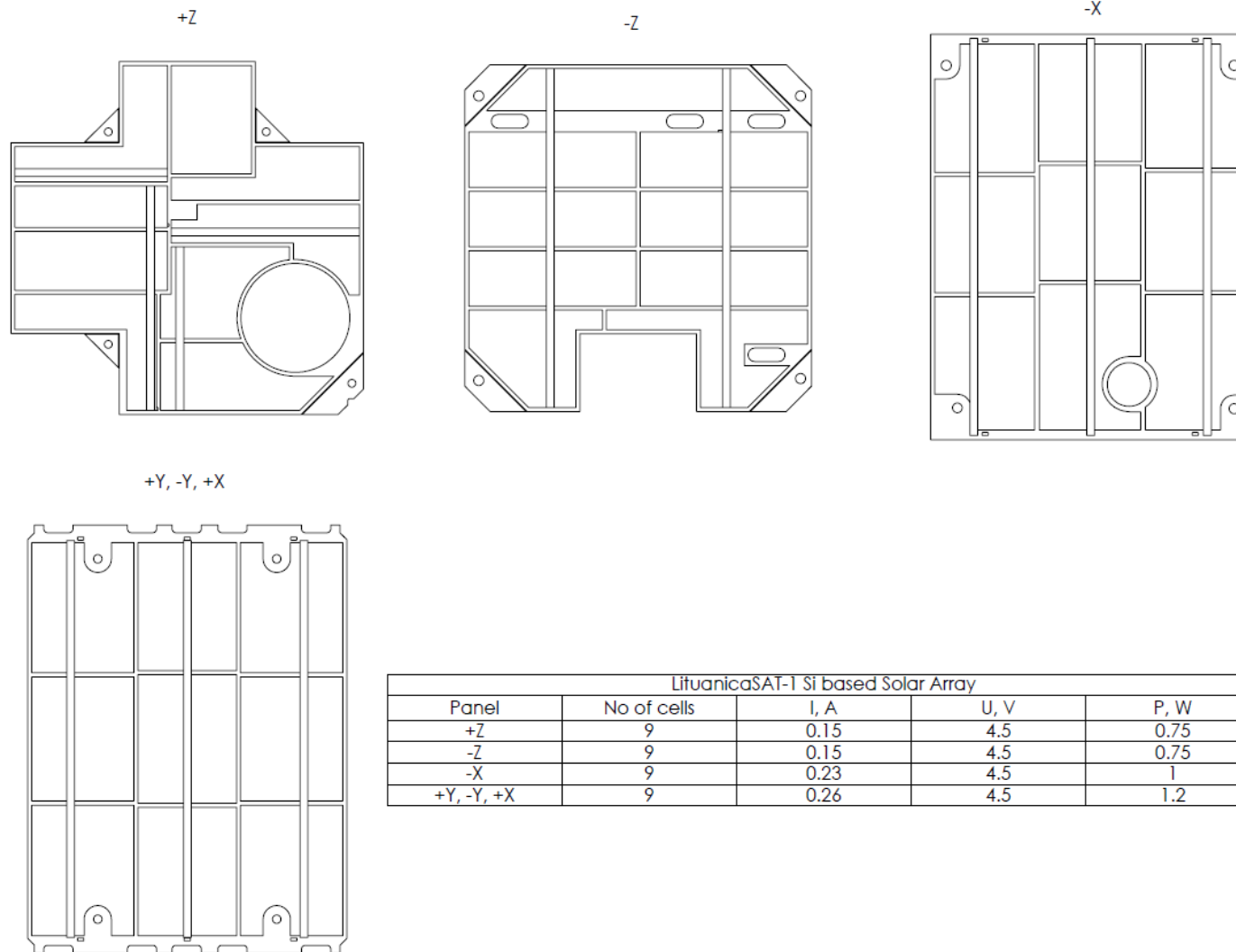
- EM – engineering model
- FM – Flight model
- CDR - *critical design review*



- GomSpace P31u as power management board
- expensive but reliable
- Design compatible with ISS safety requirements - but individual testing was needed
- Kill switch circuit had to be modified IAW NASA requirements

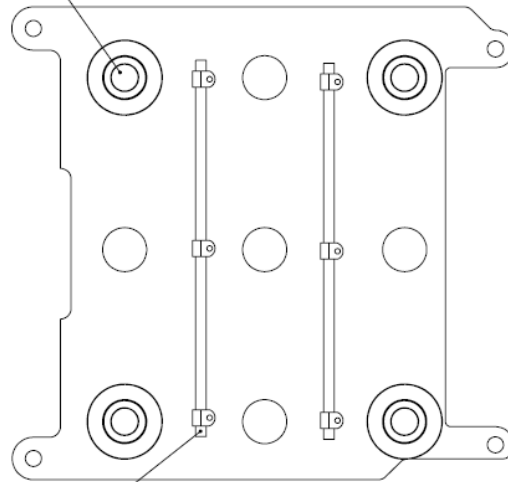


EPS – solar arrays

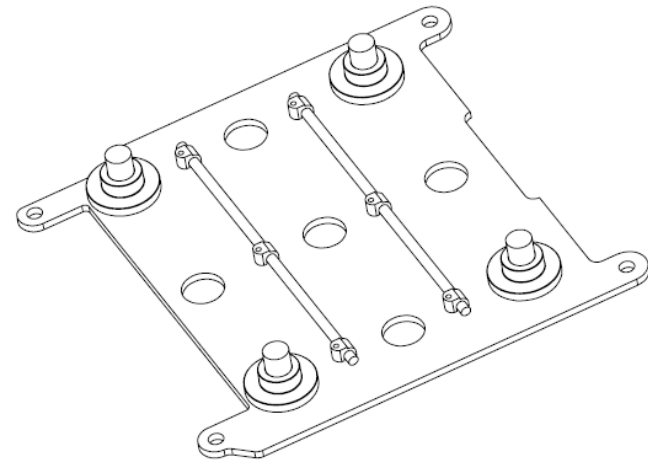


LituanicaSAT-1 Si based Solar Array				
Panel	No of cells	I, A	U, V	P, W
+Z	9	0.15	4.5	0.75
-Z	9	0.15	4.5	0.75
-X	9	0.23	4.5	1
+Y, -Y, +X	9	0.26	4.5	1.2

Permanent magnets
Alnico 5



Hysteresis material
PERMENORM 5000 H2

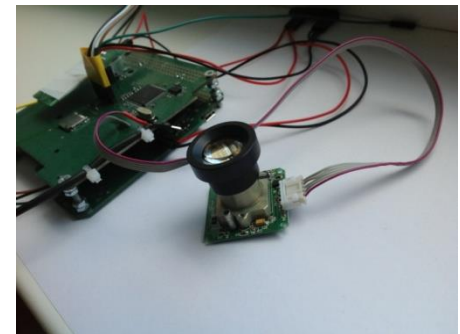


➤ Passive attitude control system:

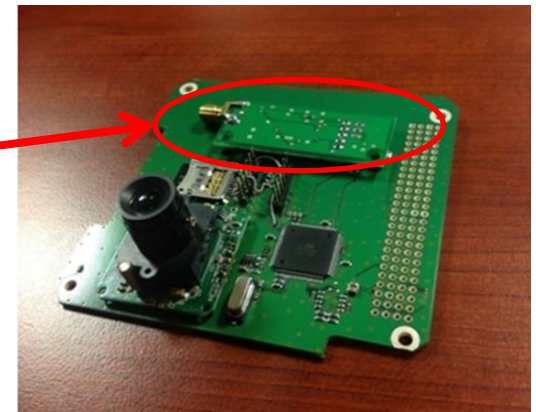
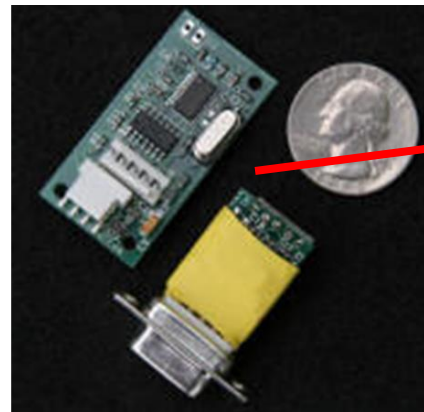
- Permanent magnet(-s): AlNiCo-5, dipole strength 0.6 A/m² (z axis)
- Hysteresis rods: PermeNorm 5000 H2, 0.075 cm³ (x and y axes)

- ARM Cortex-M4F Primary flight computer:
 - Performs power management
 - Reads and logs attitude and inertial sensor data
 - Receives and interprets telecommands from main communications transceiver
 - Sends telemetry to main communications transceiver

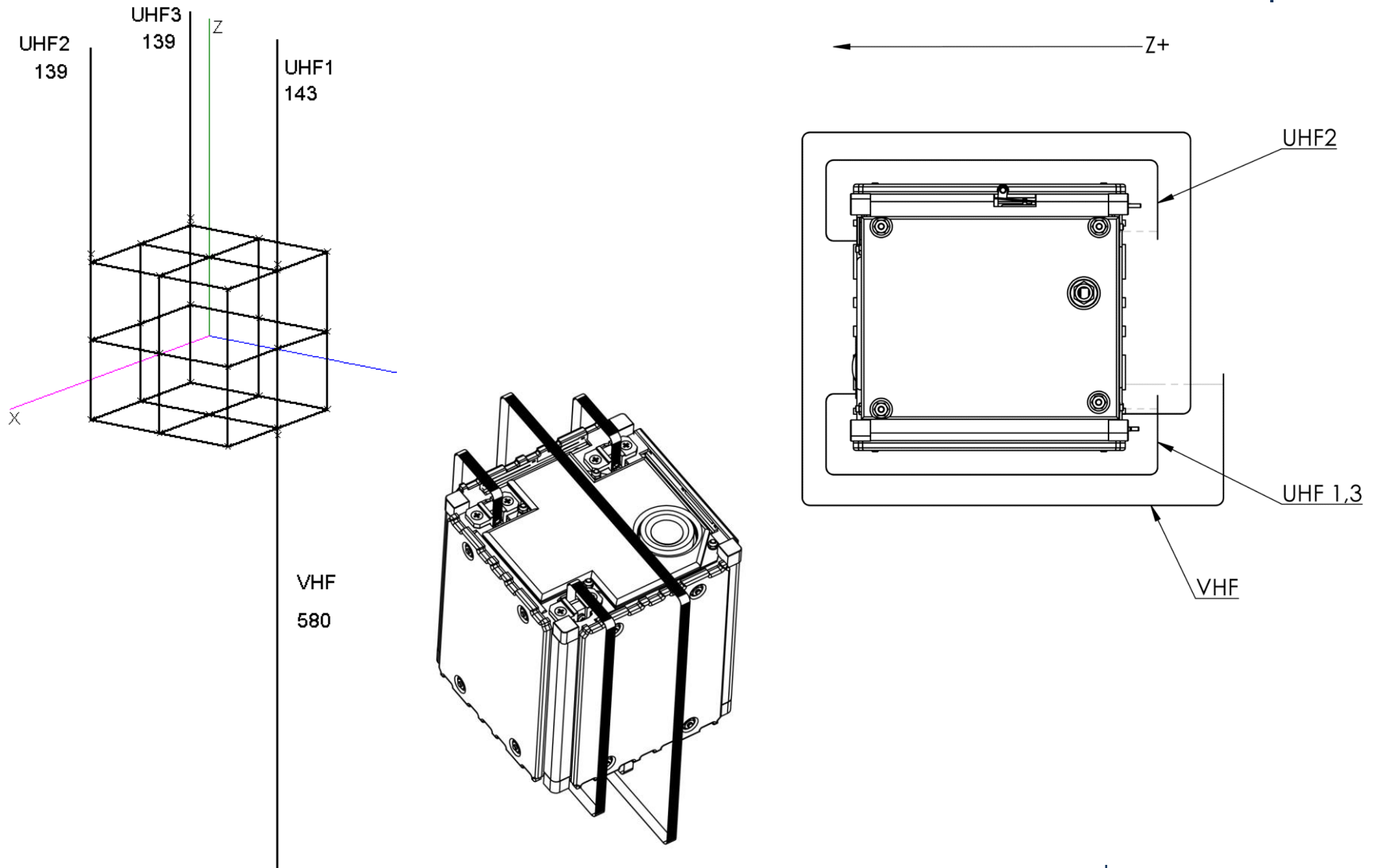
- Arduino ATmega 2560 secondary computer:
 - Controls the On board camera
 - Stores pictures in SD card flash memory
 - Controls the radio beacon
 - Routes EPS housekeeping data to the radio beacon



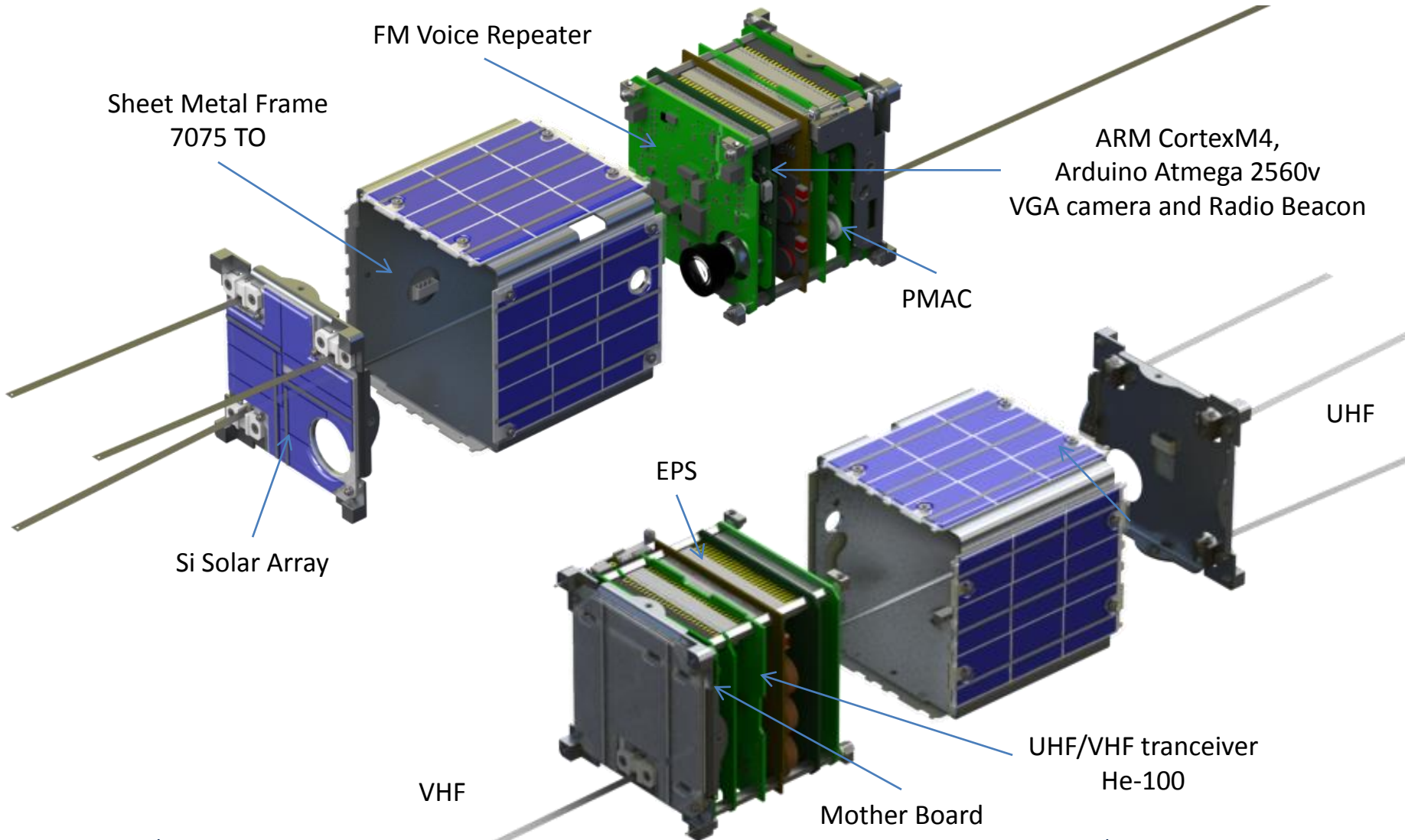
- COTS FSK/GMSK transceiver Helium-100,
 - 9k6 baud
 - 2W RF output power
- FM beacon – “Big Red Bee”
 - 100 mW output RF power
 - Flight heritage on ITupSAT-1 (>4 years in orbit)



Antenna challenge



Structure



➤ Main requirements:

Low mass, low power consumption, image processing, convenient interface, compatibility with Arduino controllers.

Selected option is LinkSprite JPEG Color Camera:

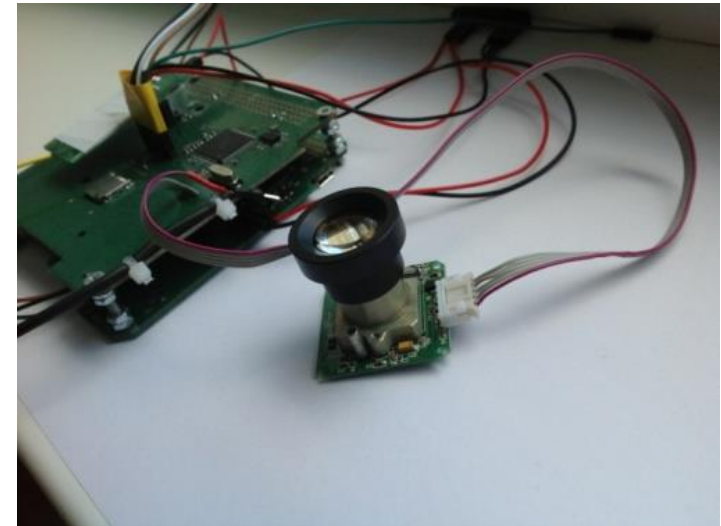
- JPEG image compression
- 3,3 - 5V power supply
- Size 32X32mm
- Current consumption: 80-100mA



Considered options:

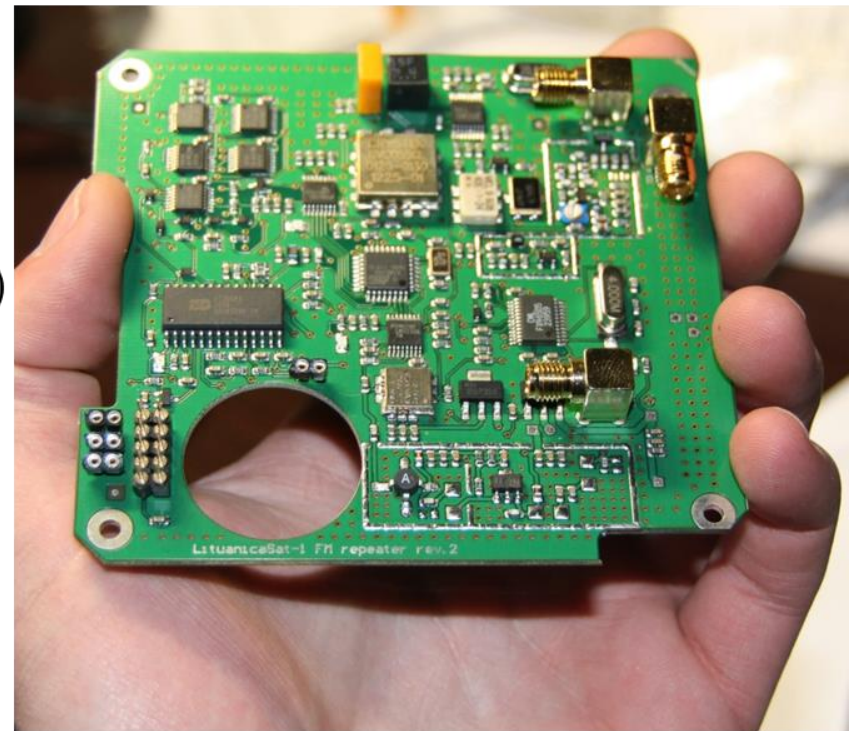
- The GomSpace NanoCam C1U
- PX4FLOW Smart Camera*

Other options were mostly rejected due to unavailability of Convenient interface and image compression possibility

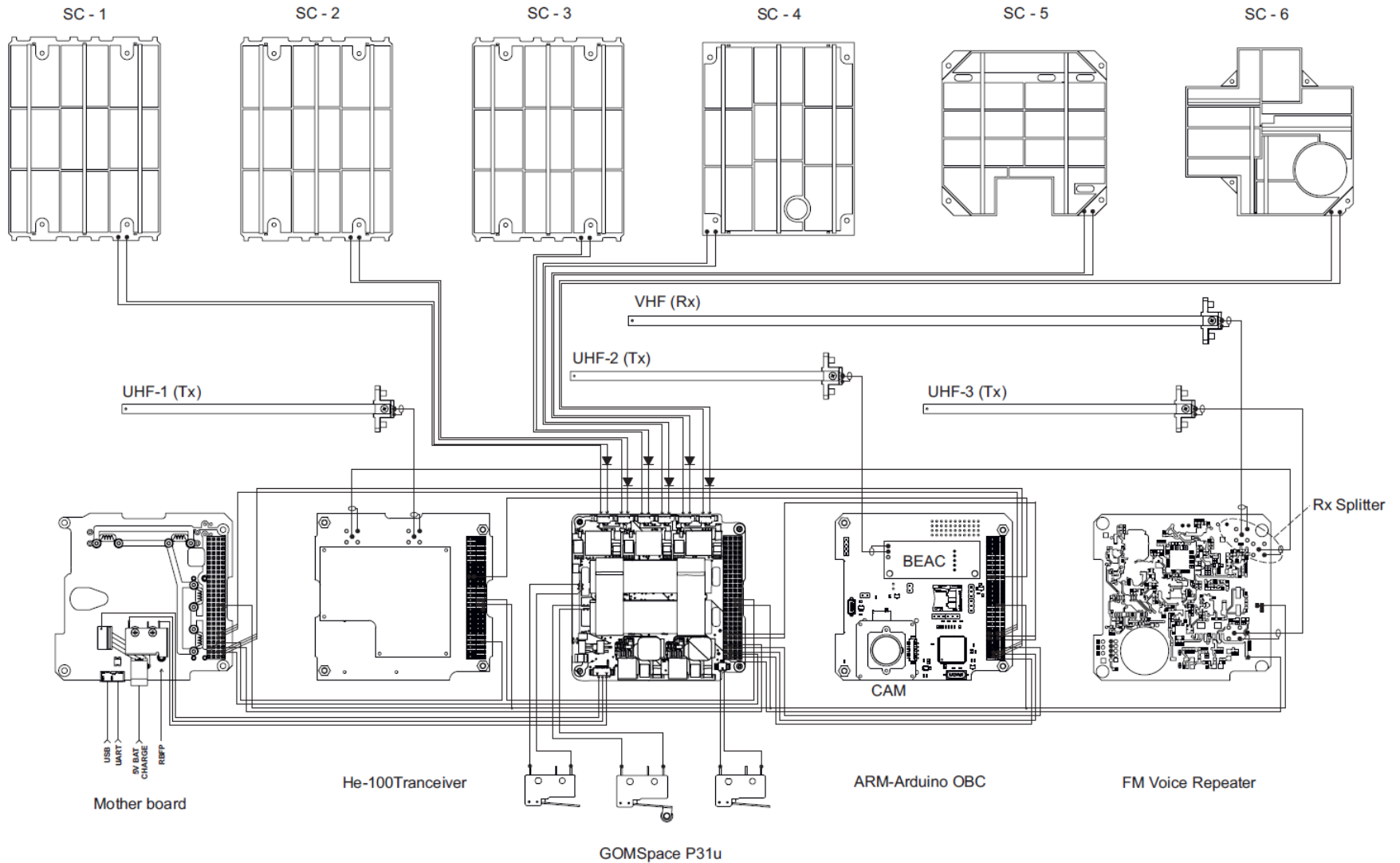


Payload – FM transponder

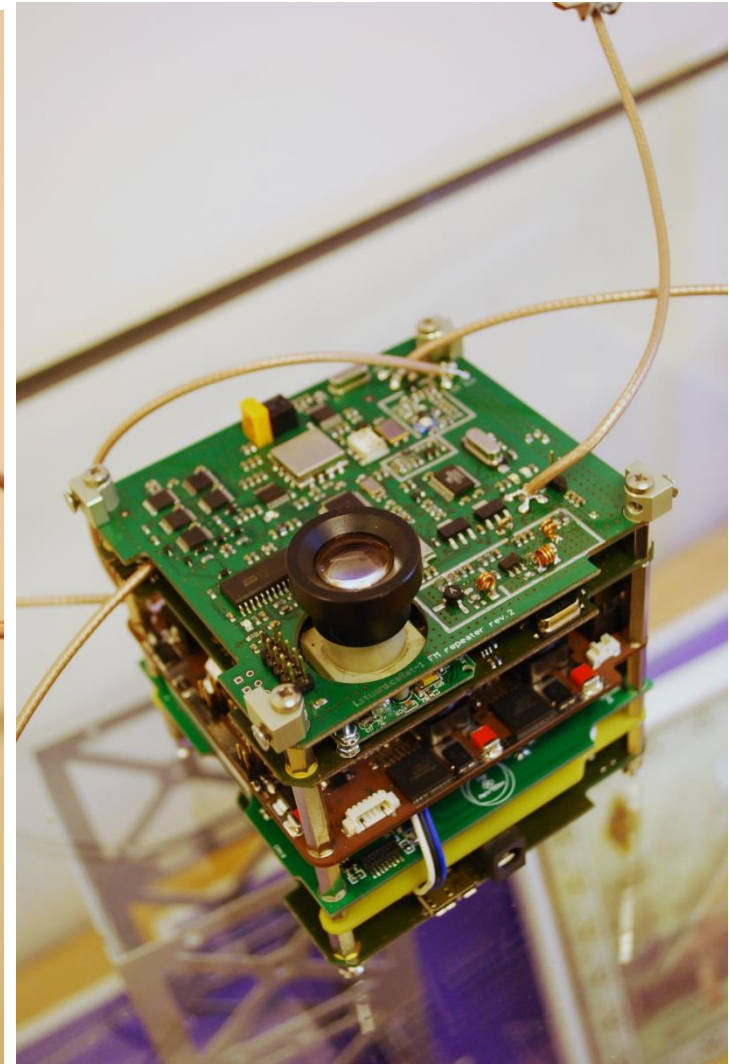
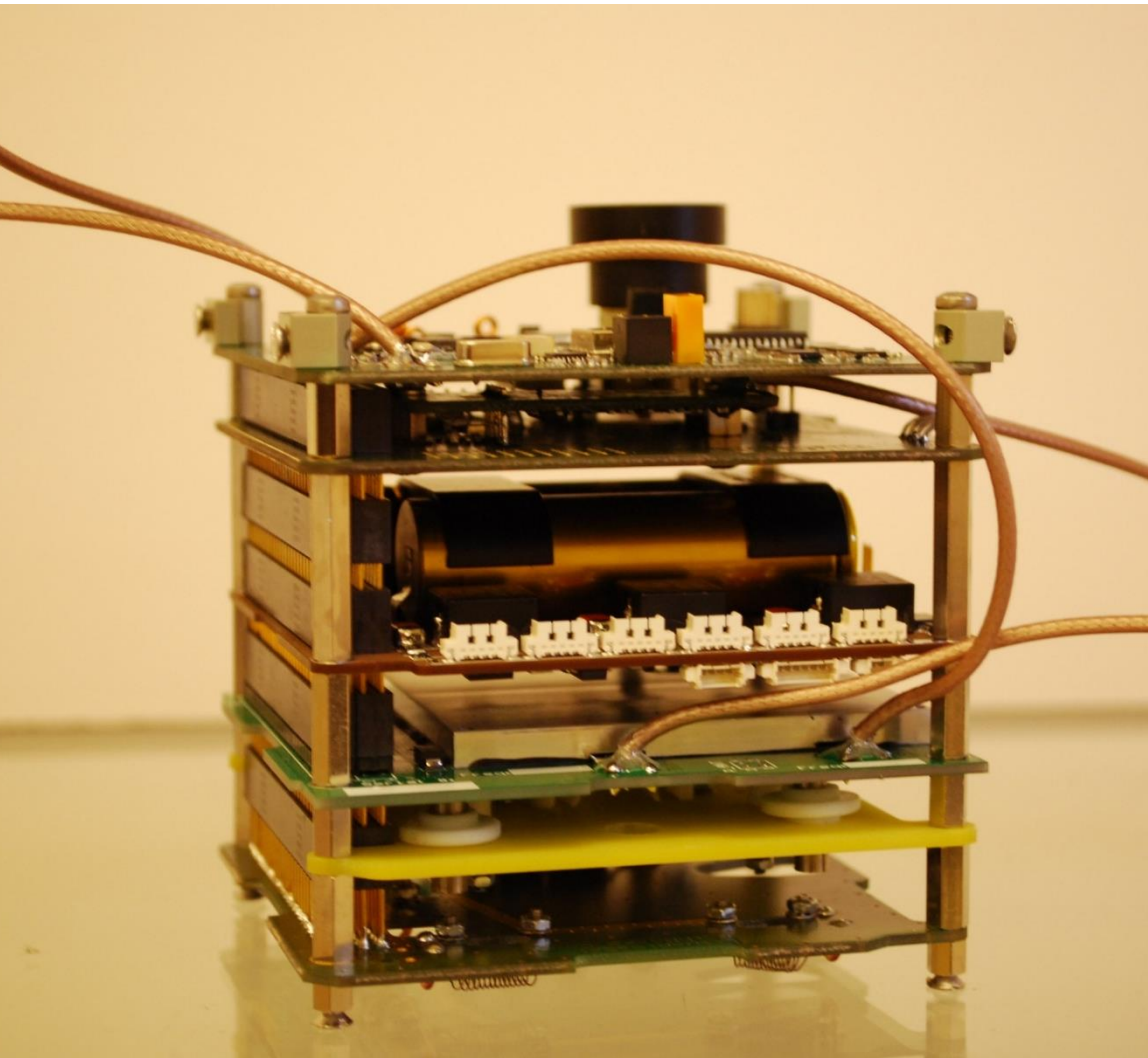
- The FM repeater is designed for placement in low Earth orbit 1U Cubesat satellite.
- The main purpose is to widen the area for amateur radio communications
- Operating temperature: -20°C to 70°C
- Weight (approx): 70 g
- Receiver system:
 - Dual conversion superheterodyne
 - 1st IF: 21.4 MHz, 2nd IF: 455 KHz
 - Frequency: 145.800-145.999 MHz
 - Receive sensitivity 0.18 μV (12 dB SINAD)
 - Selectivity: 12kHz
- Transmitter system:
 - Frequency: 435.000-438.000 MHz
 - RF Power Output: 150 mW
 - Modulation: F3E (FM)
 - Max deviation: ± 6 kHz
 - Spurious emission: -60dB
- Activation: 67 Hz CTCSS tone



LituanicaSAT-1 Electric Diagram



Protoflight model integration



Mission Operations



2014 01 09 20:07 EET

Start of the International Space Station commercial resupply mission ORB-1 (rocket carrier Antares, cargo spaceship Cygnus 2, launch pad 0Z Wallops island Virginia, USA)



Photo Credit: V. Buzas

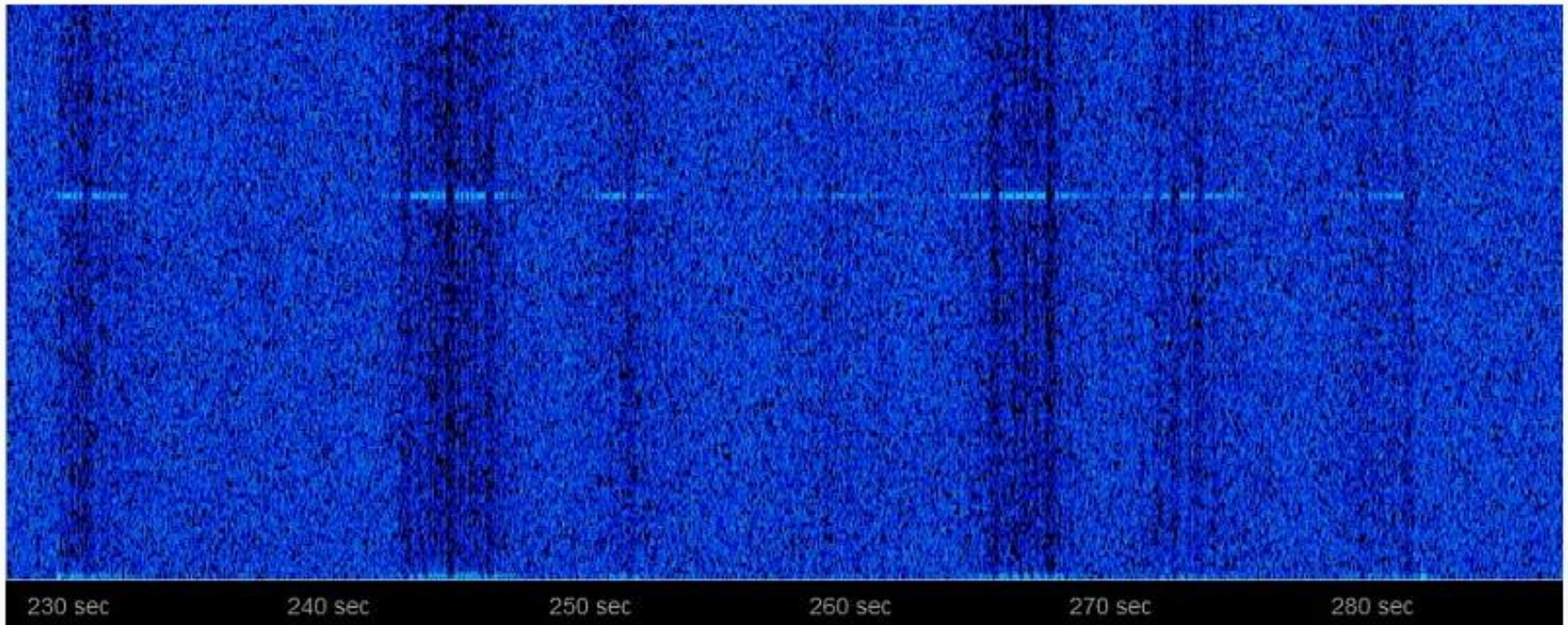
Deployment from ISS: 2014 02 28 7:30 UTC

Photo Credit: NASA



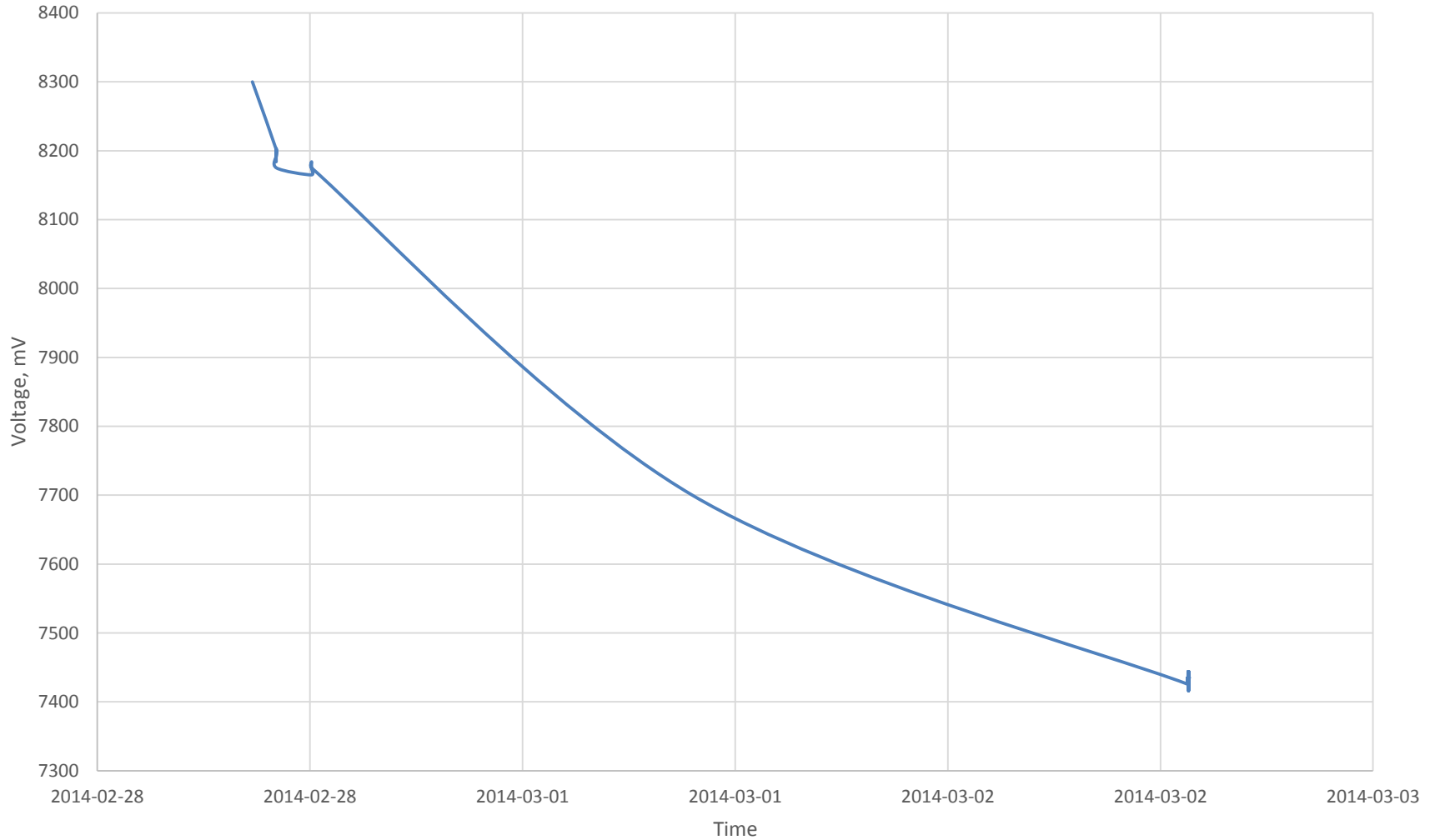
First orbit telemetry

FM beacon signal heard by DK3WN on 28/02/2014 08:45:00 UTC, 3 deg elevation pass:
LY5N.....83 C021 0112 TN05 S63



Early power problems

Battery Voltage



- Abnormally low photo current from solar panels identified as the root cause of the failure
- Solar panel voltages appeared to be nominal
- EPS under voltage protection prevented from critical failure but did not solve the problem

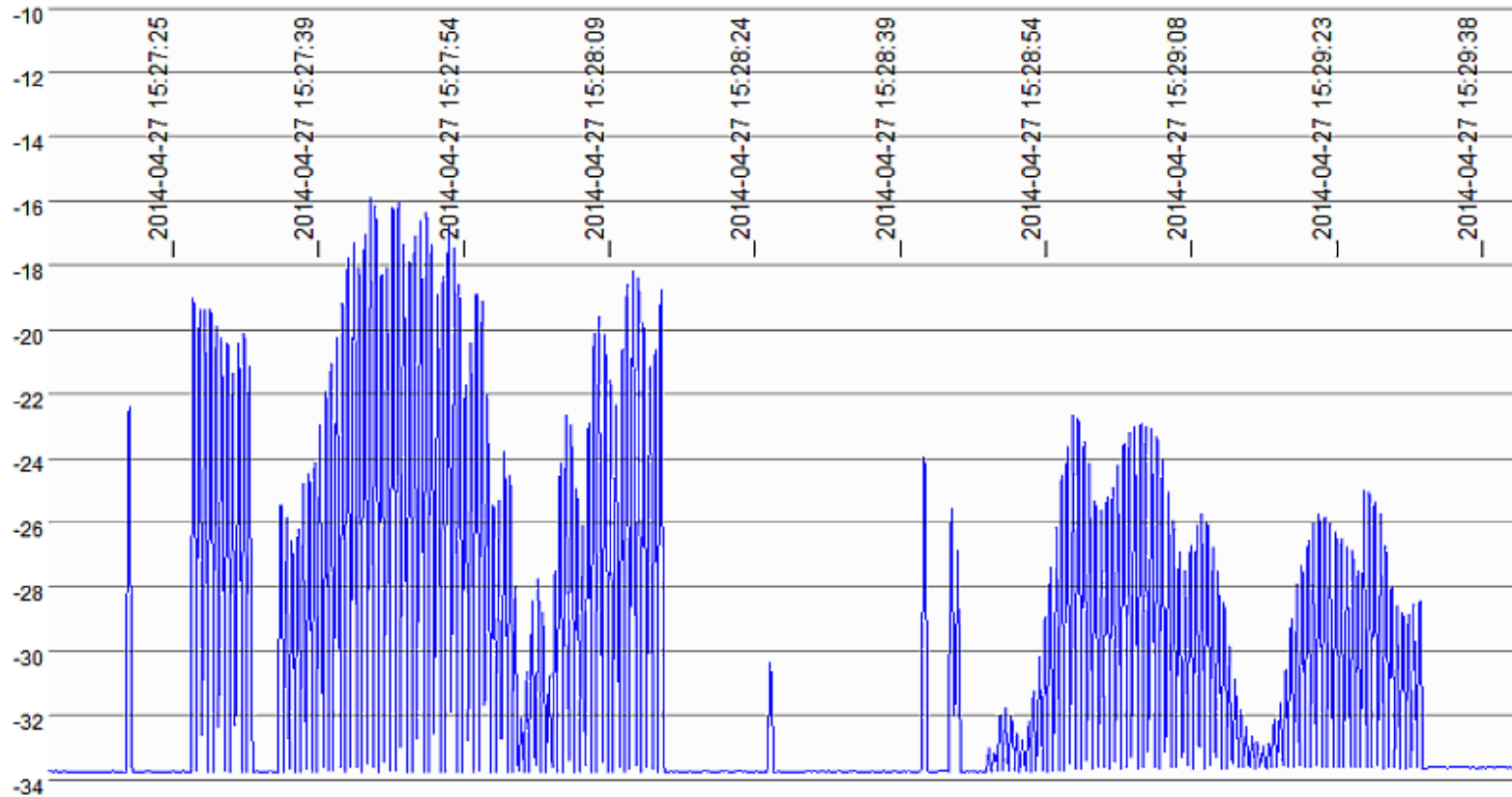


Software level setpoints	Voltage level (V)
Vnormal	7.4
Vsafe	7.2
Vcritical	6.5

- Solution: turn off the FM beacon (proved to be not so simple as it sounds)

COM troubleshooting

- Uplink budget proved to be less favorable than estimated during design (reliable uplinks were possible only using RF amplifier with 100W of output power)
- Downlink and uplink instability due to tumbling and spinning



First mission success



Successful transponder test

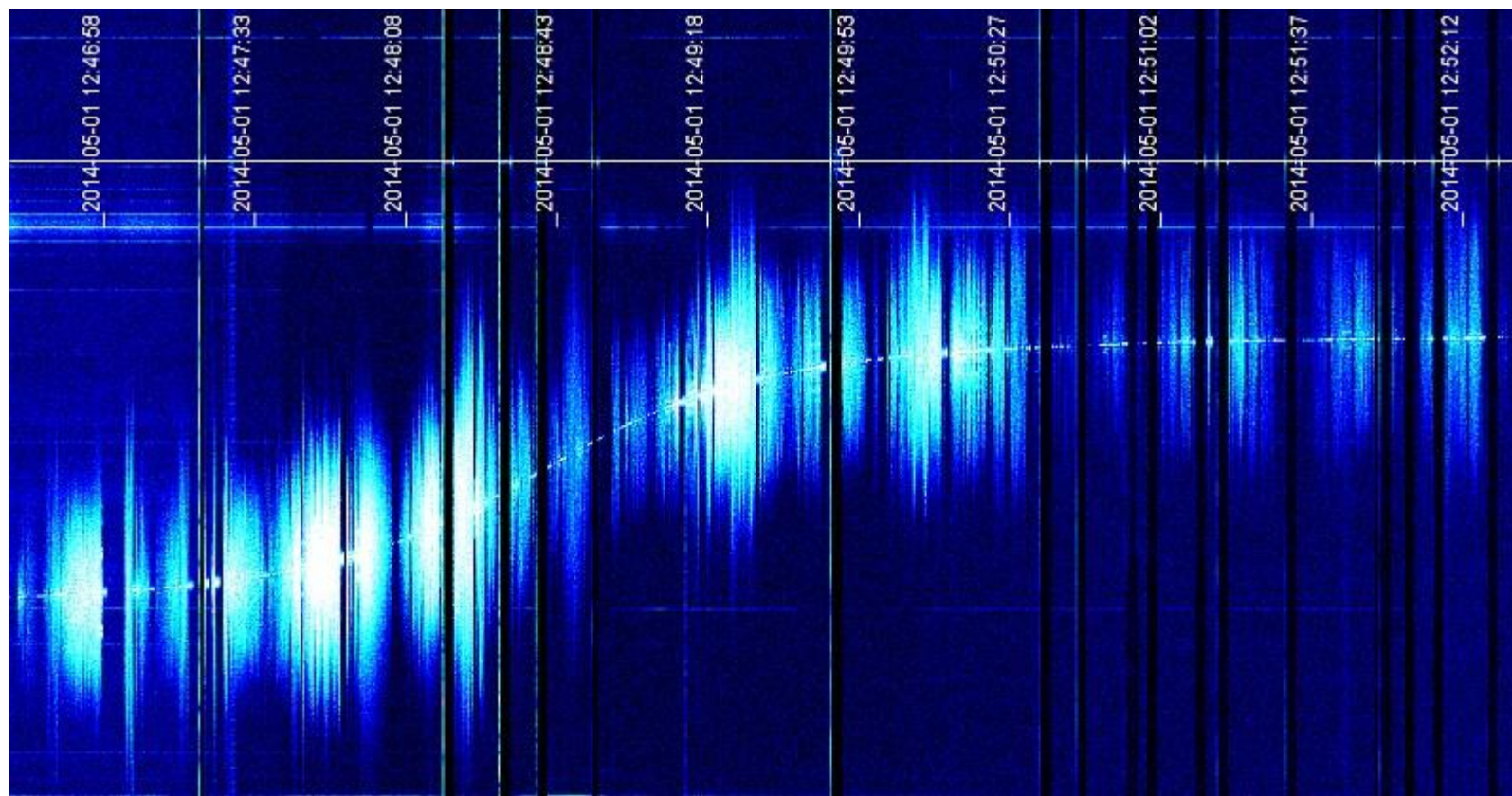
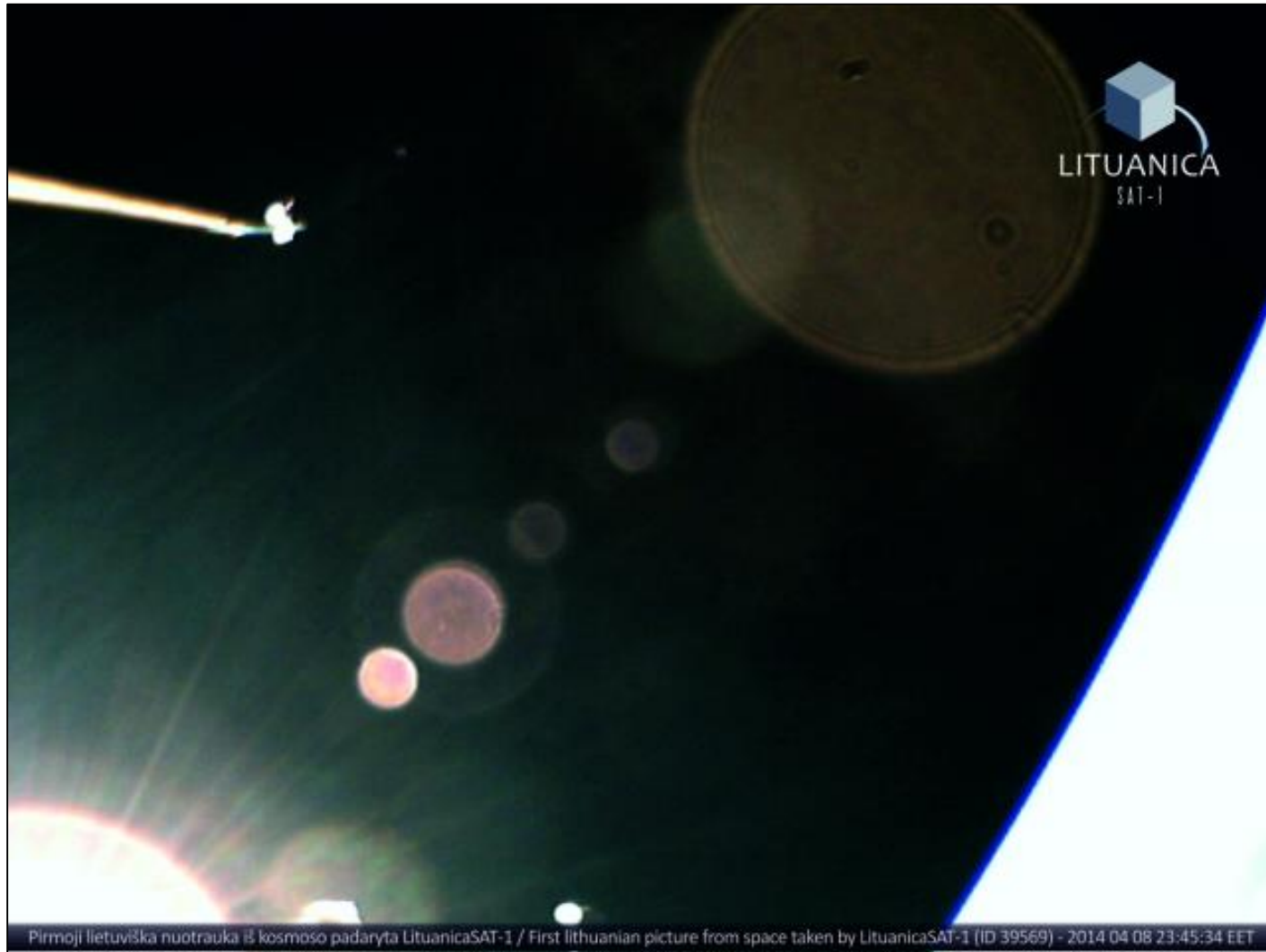


Image credit: Mike Rupprecht (url: <http://www.dk3wn.info/p/?p=44725>)

Feedback from radio amateurs



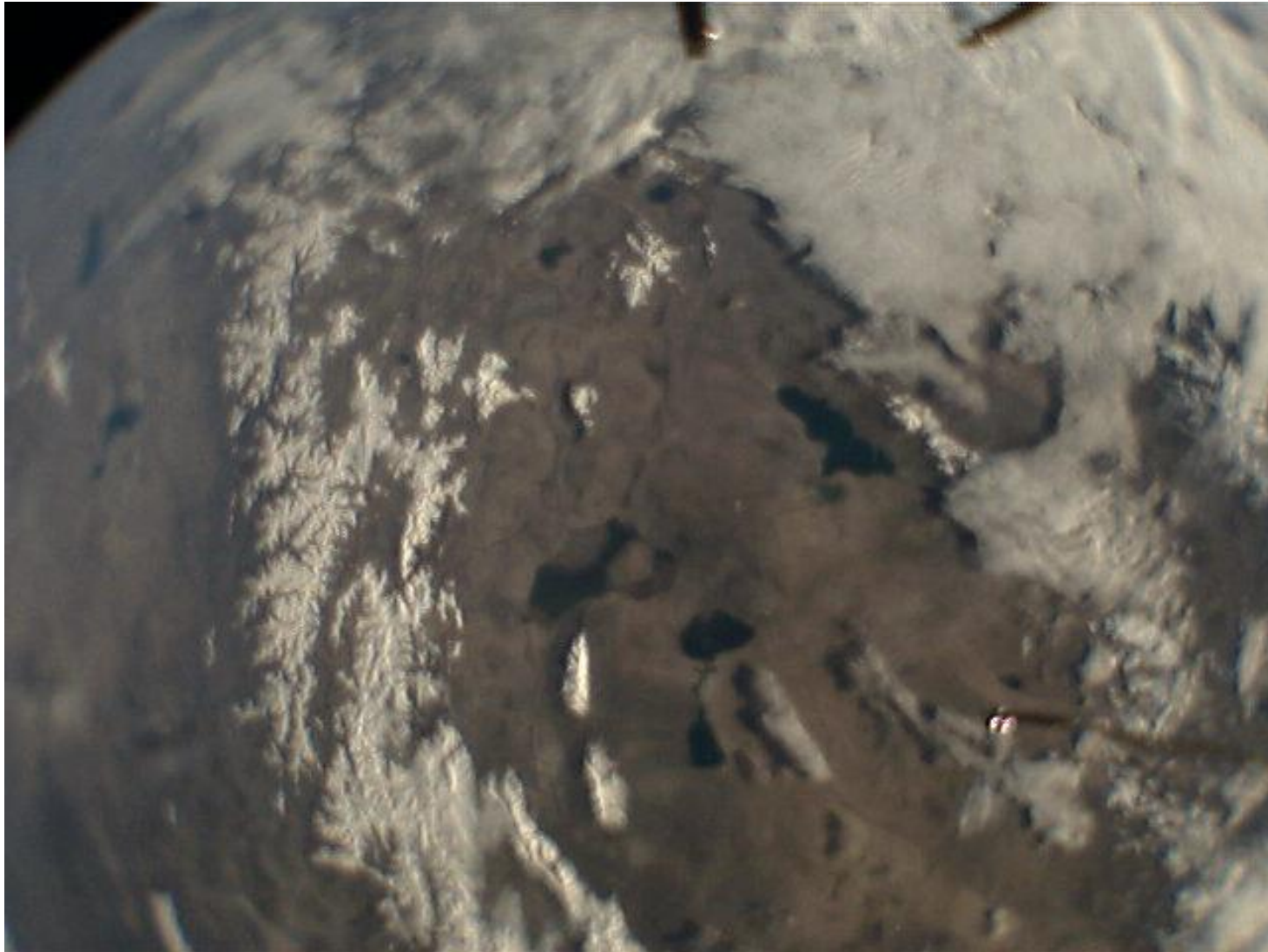
First picture from space



First glimpse at the Earth



Spying over Mongolia

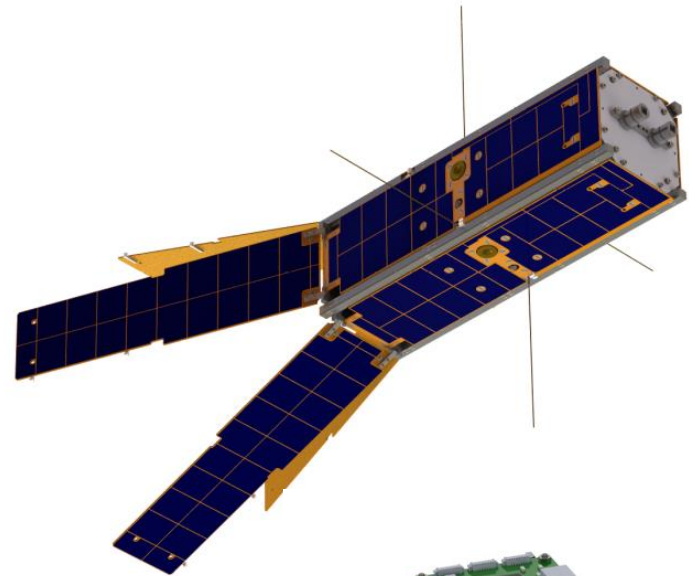


Lessons Learned

- Understand your stakeholder expectations
- Be prepared for budget overruns
- Avoid Protoflight philosophy
 - Risky (if something goes wrong – you are in a big trouble)
 - Impossible to perform troubleshooting after launch
- Software integration and testing will take more time than you expect
- Most of the problems in orbit can be avoided by thorough testing on ground
- Keep your documentation up to date and use it

Mission Objectives:

- To perform international science mission exploring lower thermosphere;
- To find new scientific and industrial partners for future collaboration;
- Technology demonstration of green propulsion system for CubeSat ($\Delta V = 150 \text{ m/s}$);
- To advance space science and technology in Lithuania;
- To develop commercial space grade nanosatellite components.



Project status:

- Flight Accepted;
- Detailed design completed and budget secured;
- Currently in phase C/D

LituanicaSAT-2 visualization (Above)
CubeSat propulsion system to be tested on LituanicaSAT-2 (Below)

Your questions

www.FB.com/LituanicaSAT-1
www.kosmonautai.lt

